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Tuesday, June 5, 2018

10 a.m. PDT • Noon CDT • 1 p.m. EDT
6 p.m. BST • 7 p.m. CEST

WEBINAR AND VIRTUAL WORKSHOP:
**ADVANCEMENTS IN GNSS+INS
TECHNOLOGY AND INTEGRATION**



WELCOME TO

Webinar and Virtual Workshop: Advancements in GNSS+INS Technology and Integration



Demoz Gebre-Egziabher
Professor
Aerospace Engineering and
Mechanics
University of Minnesota



Andrey Soloviev
Principal
QuNav



David Gaber
Marketing & Business
Development
Epson



Ryan Dixon
Chief Engineer, SPAN
NovAtel

Co-Moderator: Lori Dearman, Executive Webinar Producer

Who's In the Audience?

A diverse audience of over 450 professionals registered from 52 countries, representing the following industries:

19 % GNSS equipment manufacturer

18% System Integrator

15% Product/Application Designer

15% Professional User

13% Government

20% Other



Welcome from *Inside GNSS*



Richard Fischer
Publisher
Inside GNSS
Inside Unmanned Systems



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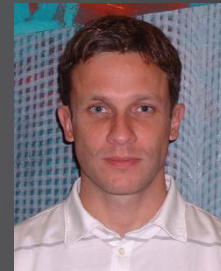
Co-Moderator: Lori Dearman, Executive Webinar Producer

Poll #1

What does having a better quality IMU improve in INS/GNSS integrated systems? (select all that apply)

- a) Accuracy*
- b) Continuity*
- c) Integrity*
- d) Availability*

GNSS/INS Integration: *Major Trends and Implementation Example*

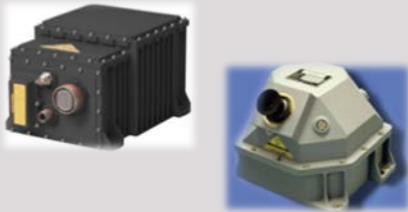


Andrey Soloviev
Principal
QuNav

Key Trends

- From high-grade to lower SWAP-C IMUs

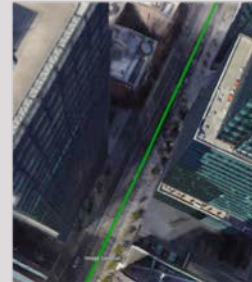
RLG, FOG



MEMS



- Application for GNSS-degraded and denied environments



Main Challenge

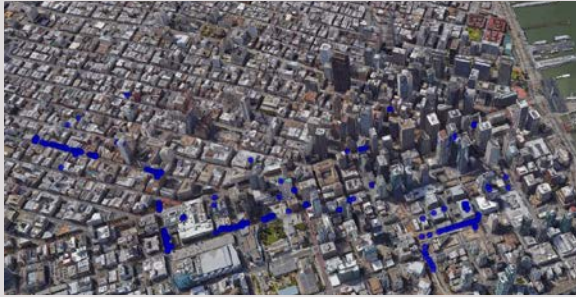
Mitigation of inertial error drift

Solutions:

- Use of *advanced integration techniques*
- *Integration with other sensors*

Advanced Integration Techniques

Loose coupling (integration at solution level) has *limited benefits*



- Only sparse GNSS fixes can be obtained;
- This results in extended GNSS outages

Tight coupling and deep integration must be used:

- **Tight coupling** (integration at the measurement level):
 - Increases the availability of (partial) GNSS updates
- **Deep integration** (integration at the signal processing level):
 - Weak signal recovery
 - Multipath suppression

Integration with Other Sensors

- GNSS/INS performance can be still limited
- Integration with other sensors (and sources of navigation data) must be used:

Examples:

Video-cameras;

Motion models (non-holonomic constraints)

Integration Example:

GNSS/INS for consumer-grade IMUs

Main Features of Consumer-Grade Inertial



Key challenges:

- Large sensor errors
- Partially defined (undefined) specs (e.g., axes misalignment)
- Nonlinearities (heading drift)

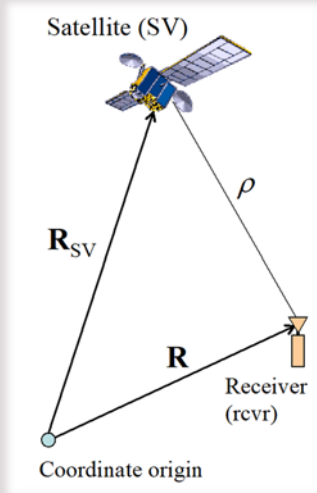
However:

- Bias (drift) stability and noise performance has improved significantly;
- This enables the use of **consumer-grade IMUs** for **improved robustness of GNSS** (coasting through outages and weak signal recovery)

Use of GNSS Carrier Measurements

- Large (but stable) biases are still present: e.g. gyro drift at a 5 deg/s level
- GNSS carrier phase (or Doppler frequency) can be utilized to estimate and remove bias components: low-noise measurements enable fast convergence

Integration with GNSS Carrier Phase



$$\varphi = \rho + \lambda N + \delta t_{rcvr} + \varepsilon + \eta$$

\downarrow \downarrow \downarrow \swarrow
Integer ambiguity *Clock bias* *Atmospheric delays & SV clock* *Noise & multipath*

- **Resolving integer ambiguities** can be **challenging**:
 - Need for a base station;
 - Limited number of SVs
- Therefore, **carrier phase changes** are used as **GNSS observables**:

$$\Delta\varphi = \varphi(t_n) - \varphi(t_{n-1}) = \Delta\rho + \Delta\delta t_{rcvr} + \Delta\varepsilon + \Delta\eta$$

\downarrow
Directly related to INS error states

INS Error Model

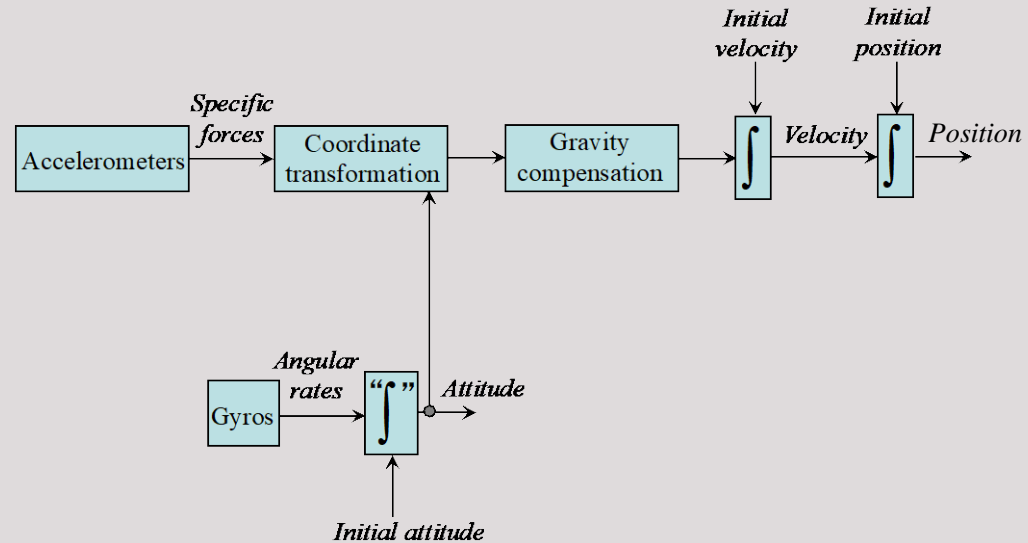
24 states:

- Position errors (3 states)
- Delta position errors (3 states)
- Velocity errors (3 states)
- Attitude errors (3 states)
- Gyro and accelerometer biases (6 states)
- Axis misalignment (6 states)

INS Navigation Mechanization

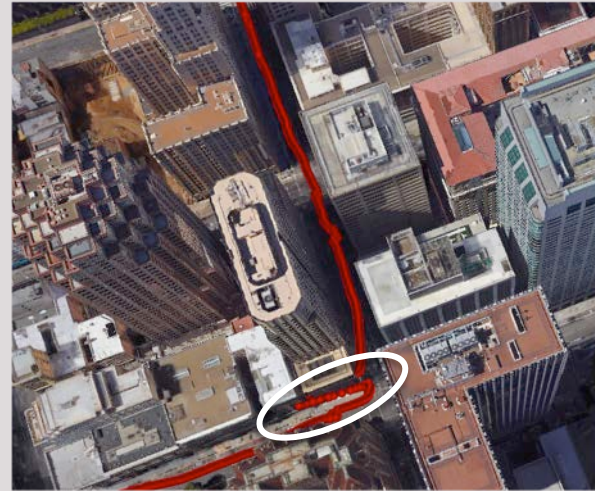
Relatively simple mechanization can be used:

- No need to compensate for non-inertial effects that are below the level of sensor errors



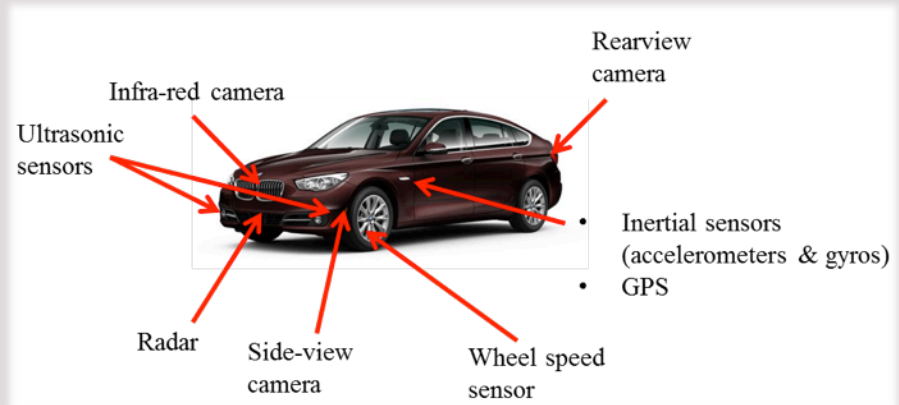
Performance of GNSS/INS integration can be still limited...

- Performance of GNSS-only solution is improved significantly
- However, some limitations remain: e.g., large position errors (tens of meters) can be present in urban canyons



Improving performance of GNSS/INS

- Other sources of navigation information have to be used for reliable navigation
- Vehicle motion model: non-homonymic constraints
- Integration with other sensors

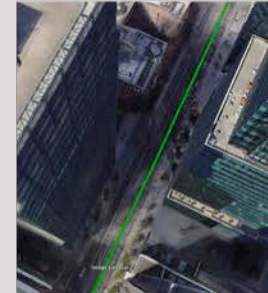
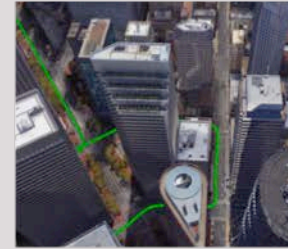


Example performance in urban canyons (downtown San Francisco)

Carrier phase GNSS/INS (STMicro iNEMO)/motion model/monocular video camera



Reliable positioning is maintained for the entire duration of the test

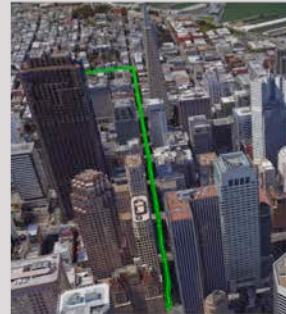


Example performance in urban canyons (downtown San Francisco)

Carrier phase GNSS/INS (STMicro iNEMO)/motion model/monocular video camera

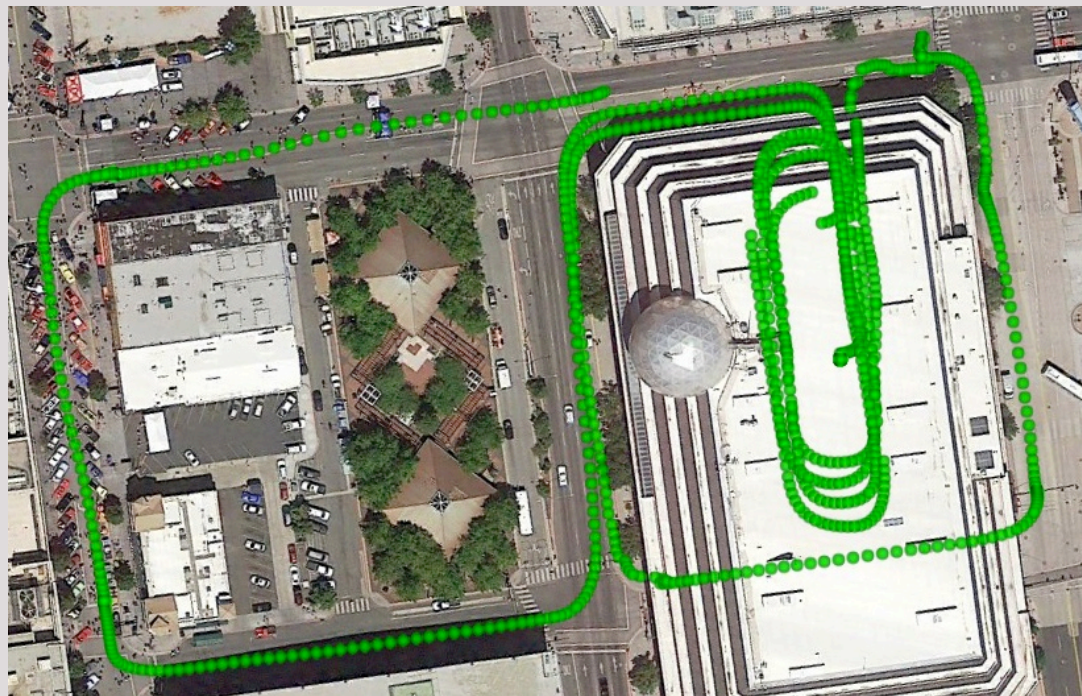


Reliable positioning is maintained for the entire duration of the test



Example performance in a parking garage

- GNSS/INS/motion model
- Consumer-grade IMU (STMicro iNEMO)
- GNSS outage duration exceeds 5 min

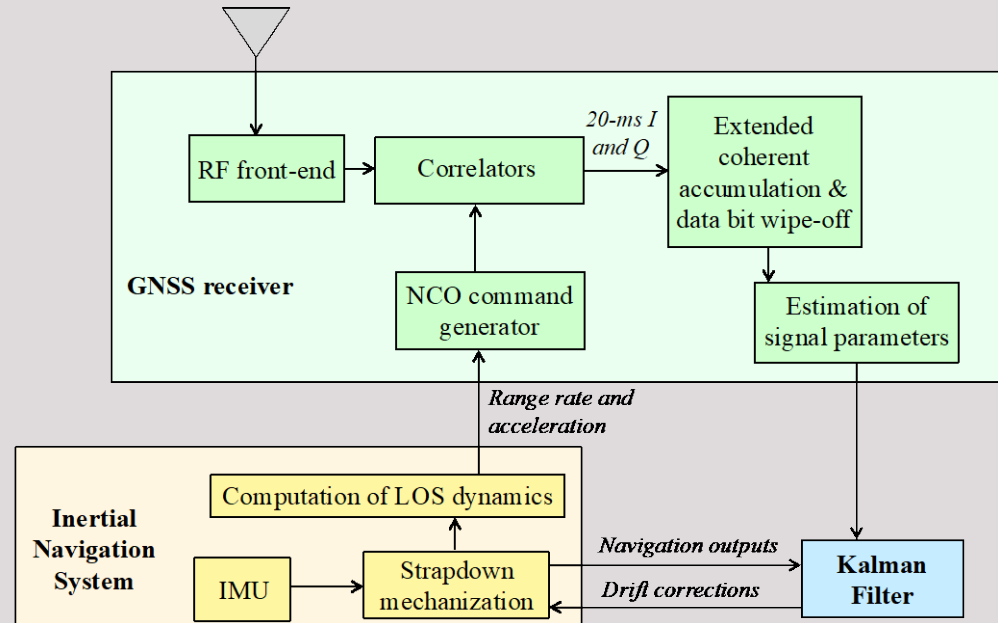


Use of consumer-grade IMU to enhance robustness of GNSS signal processing

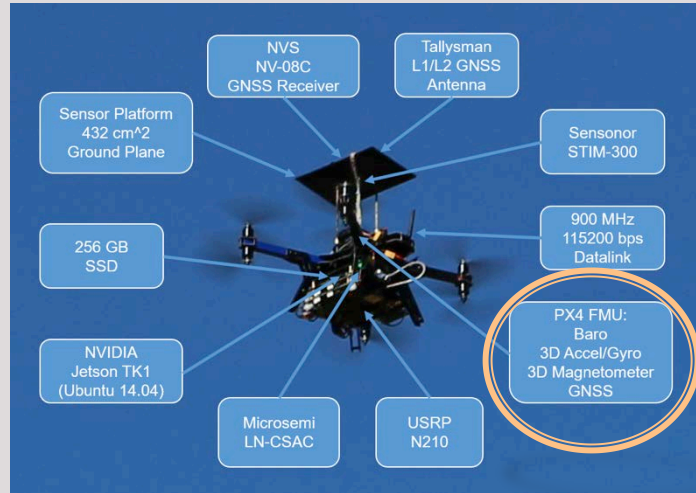
- Consumer-grade IMU is used to extend coherent accumulation of GNSS signals
- This enables:
 - Weak signal recovery (thus enhancing the GNSS signal availability)
 - Multipath suppression

Deep GNSS/INS integration with long coherent integration

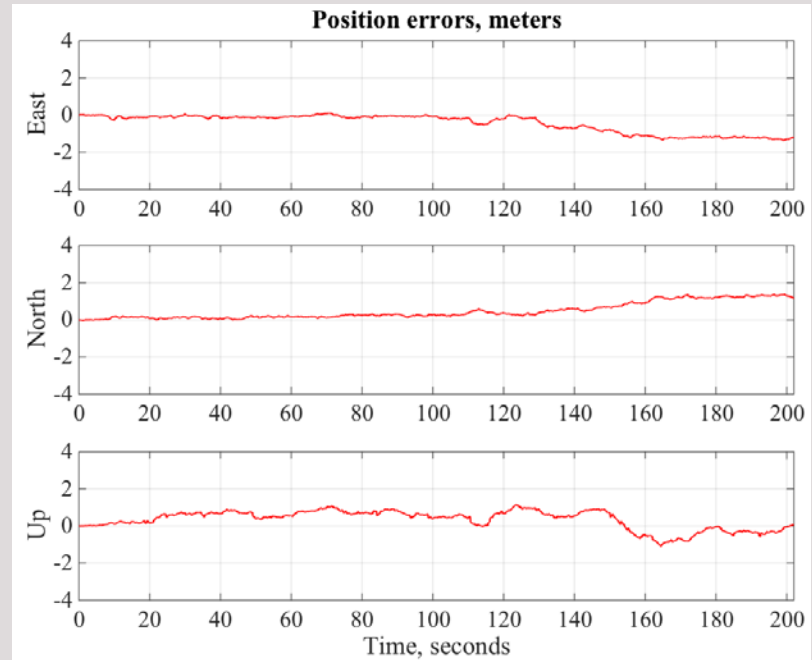
- Traditionally, deep integration has been developed for navigation grade and/or tactical-grade IMUs;
- Improved performance of consumer-grade MEMS IMUs allows for deep integration with low-cost inertial sensors



UAV demonstration example



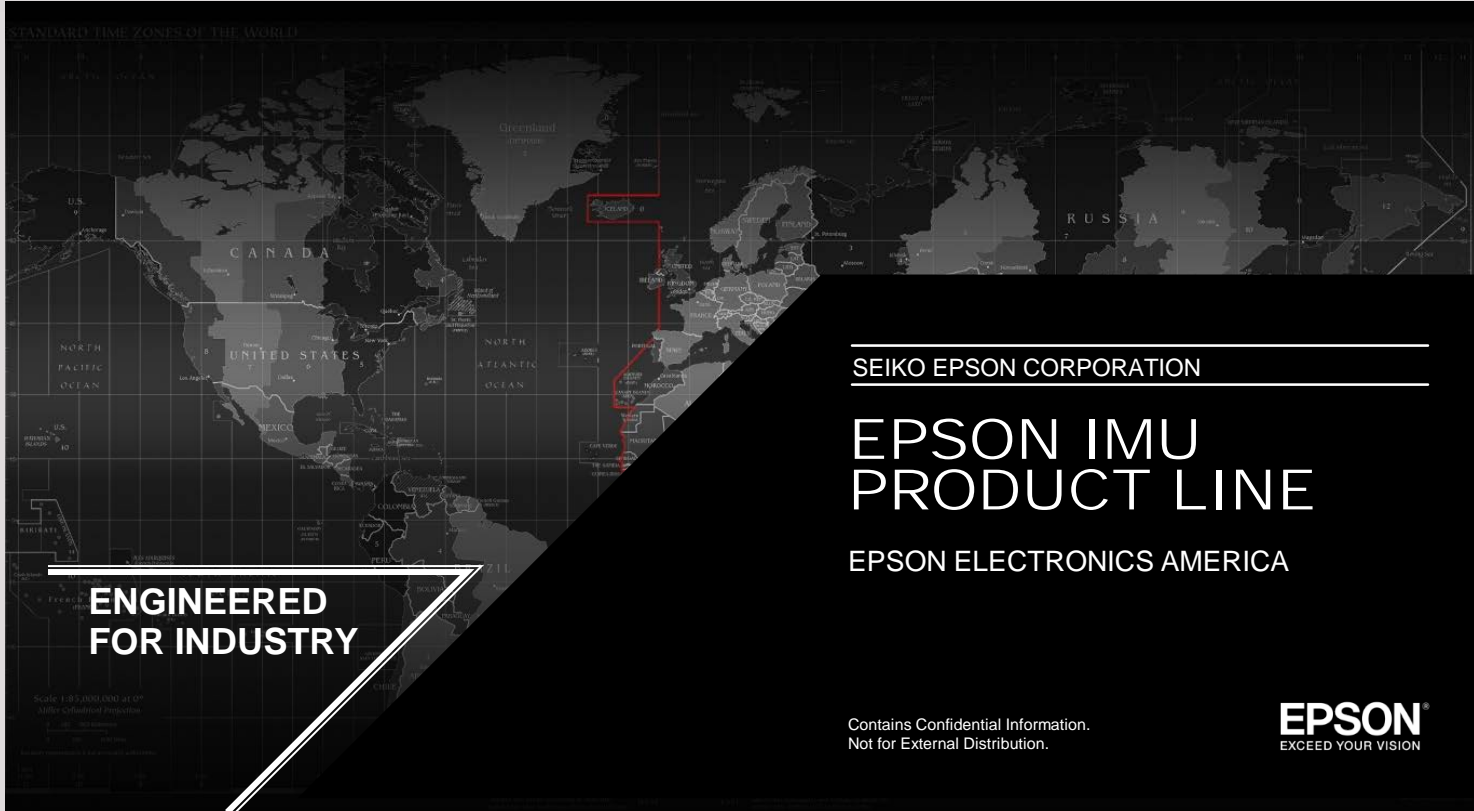
Position performance under 40-dB attenuation introduced into live-sky GPS signals



EPSON IMU FEATURE SET



David Gaber
Marketing & Business
Development
Epson



STANDARD TIME ZONES OF THE WORLD

**ENGINEERED
FOR INDUSTRY**

Scale: 1:9,500,000 at 0°
After Continued Projection

SEIKO EPSON CORPORATION

**EPSON IMU
PRODUCT LINE**

EPSON ELECTRONICS AMERICA

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OVERVIEW

GYRO TECHNOLOGY
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CURRENT PRODUCTS
NEW PRODUCTS

Epson IMU History

- Vertical Integration
- Quartz Crystal
- QMEMS
- QMEMS Uses

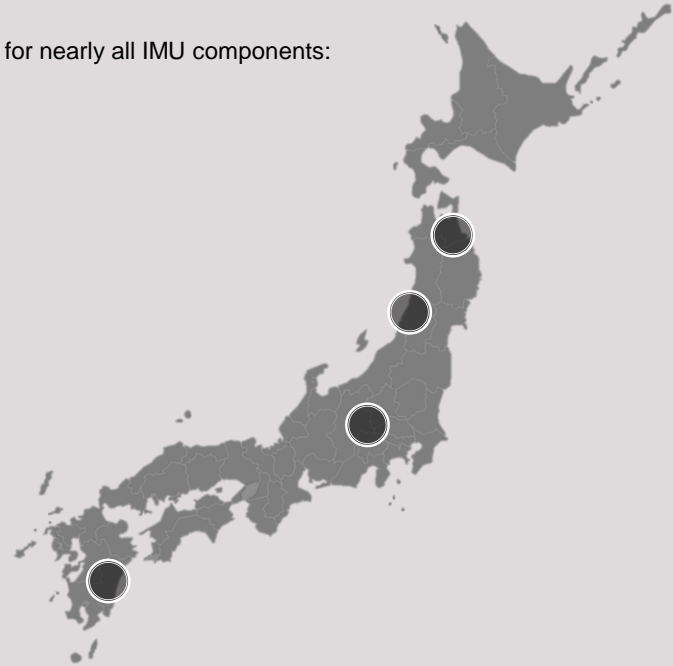
SECTION I: OVERVIEW

EPSON IMU HISTORY

Vertical Integration

Epson owns and controls all manufacturing and production for nearly all IMU components:

- Synthetic Crystal Bar Production
 - Hachinohe Plant, Japan
 - Miyazaki Plant, Japan
 - Washington Plant, USA
- Wafer Processing
 - Hachinohe Plant, Japan
- IC & MCU Fabrication
 - Sakata Plant, Japan
 - Fujimi Plant, Japan
 - Suwa Minami Plant, Japan
- Gyro Fabrication
 - Ina Plant, Japan
- Oscillator Fabrication
 - Miyazaki Plant, Japan
 - Shonan Plant, Japan
- Final Assembly
 - Fujimi Plant, Japan
- Testing and Calibration
 - Fujimi Plant, Japan
 - Sakata Plant, Japan



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- QMEMS Uses

SECTION I: OVERVIEW

EPSON IMU HISTORY

Quartz Crystal

Epson produces and utilizes 100% synthetic quartz crystal for all inertial sensing products:

- Epson Synthetic Quartz Crystal
 - Synthesized from natural quartz crystal
 - Uniform size, shape and quality
 - Efficient wafer yielding = low production costs
- Natural Quartz Crystal
 - Found in nature but very expensive
 - Varies in size and shape
 - Contains impurities = susceptible to cracks



OVERVIEW

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Epson IMU History

- Vertical Integration
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- QMEMS Uses

SECTION I: OVERVIEW

EPSON IMU HISTORY

QMEMS

Epson's proprietary quartz MEMS fabrication process:

- Stable Supply
 - High quality supply of synthetic quartz is available throughout Japan & United States
- Physically and Chemically Stable Material
 - Low aging = excellent long-term stability
 - Excellent workability and low variation among samples
- Extremely Low Internal Loss of Vibration
 - Low power required for oscillation = low overall power consumption
- Performance over Temperature can be Dictated by Cutting Angle
 - Proprietary cutting angle process and technology assures consistent performance



OVERVIEW

- GYRO TECHNOLOGY
- IMU EVOLUTION
- CURRENT PRODUCTS
- NEW PRODUCTS

Epson IMU History

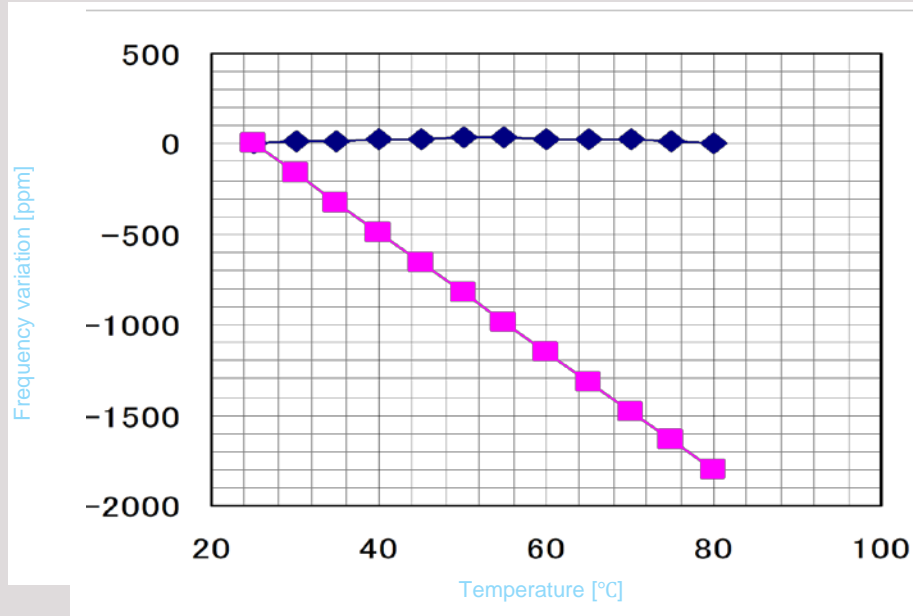
- Vertical Integration
- Quartz Crystal
- QMEMS
- QMEMS Uses

SECTION I: OVERVIEW EPSON IMU HISTORY

QMEMS

Epson's QMEMS elements are highly stable over temperature:

- QMEMS gyroscopes offer ~100x better stability than SiMEMS.



Crystal

30ppm

Silicon

3,000ppm

OVERVIEW

GYRO TECHNOLOGY
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Epson IMU History

- Vertical Integration
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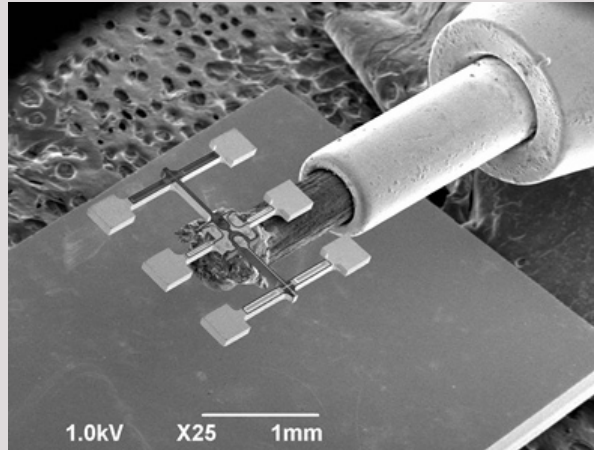
SECTION I: OVERVIEW

EPSON IMU HISTORY

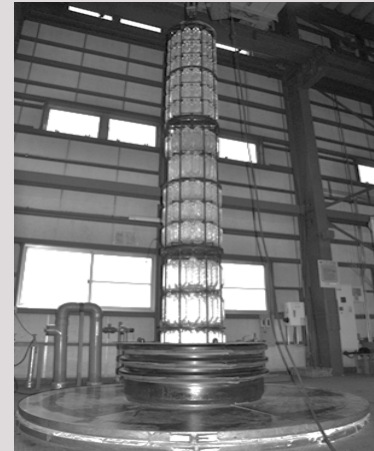
QMEMS Uses

Epson's proprietary quartz MEMS fabrication process is used for many product lines:

- Timing Products
- Real-Time Clocks
- Inertial Sensors



A QMEMS element for a gyrosopic sensor is shown balanced on the tip of a pencil lead.



One of Epson's QMEMS autoclaves located in Japan.

OVERVIEW
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Proprietary Element

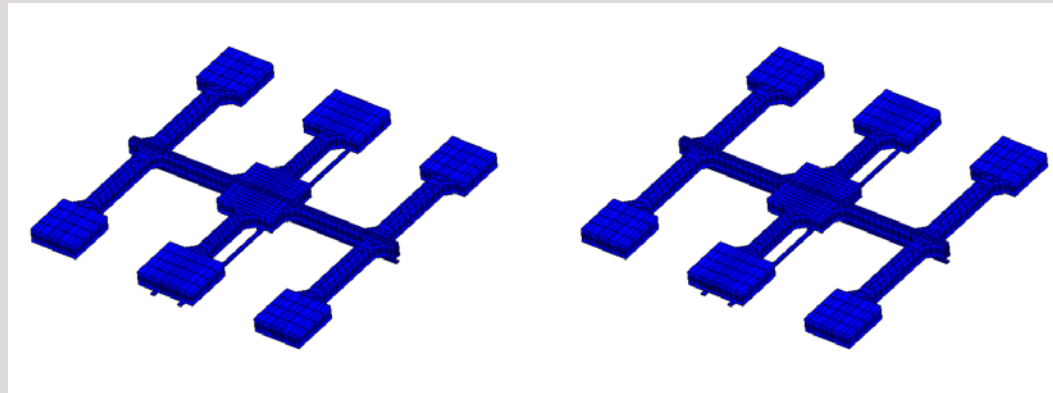
- Physical Structure
- Architecture
- Differentiation
- Vibration Effects
- Shock Effects
- Temperature Effects
- Noise Density

SECTION II: GYRO TECHNOLOGY PROPRIETARY ELEMENT

Physical Structure

Epson's proprietary "Double-T" quartz MEMS gyroscopic sensor:

- Operates like traditional Coriolis gyros
- Uses differential detection
- Drive and detection arms vibrate in the same plane



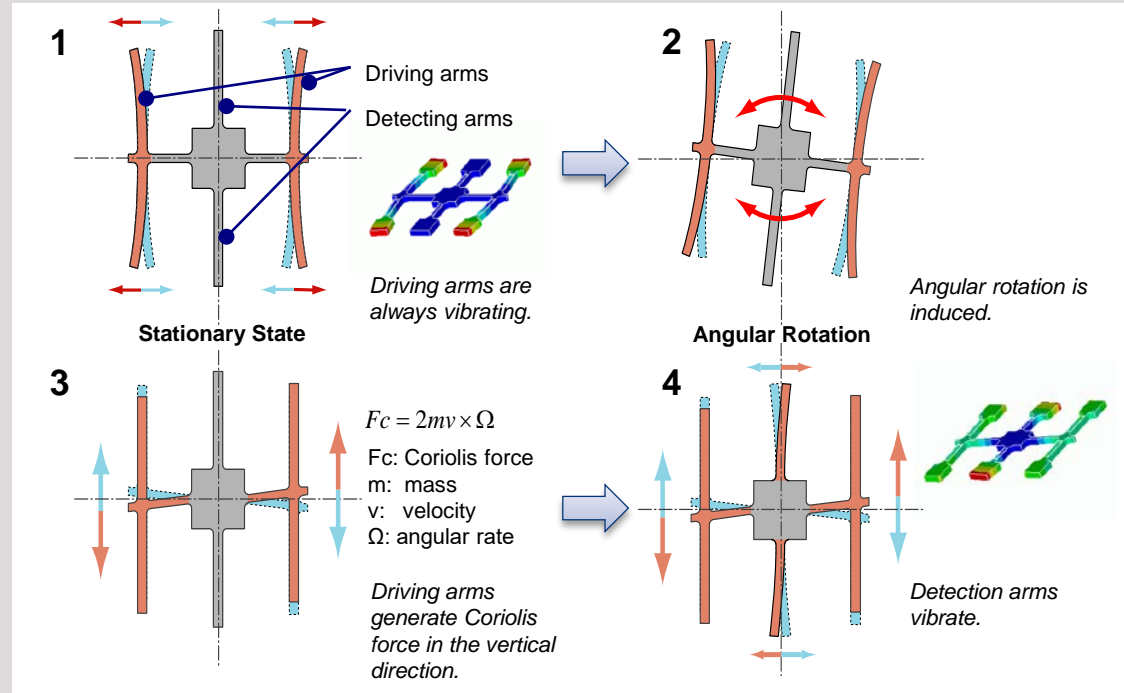
A QMEMS element shown in drive mode (left) and detection mode (right).

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Proprietary Element
 — Physical Structure
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 — Temperature Effects
 — Noise Density

SECTION II: GYRO TECHNOLOGY PROPRIETARY ELEMENT

Physical Structure
 Epson's QMEMS
 Double-T
 gyroscopic
 sensor principle
 of operation:

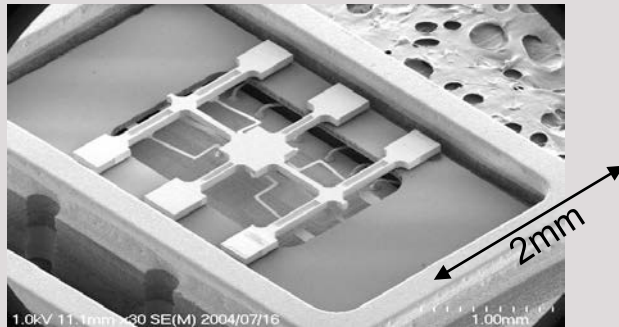
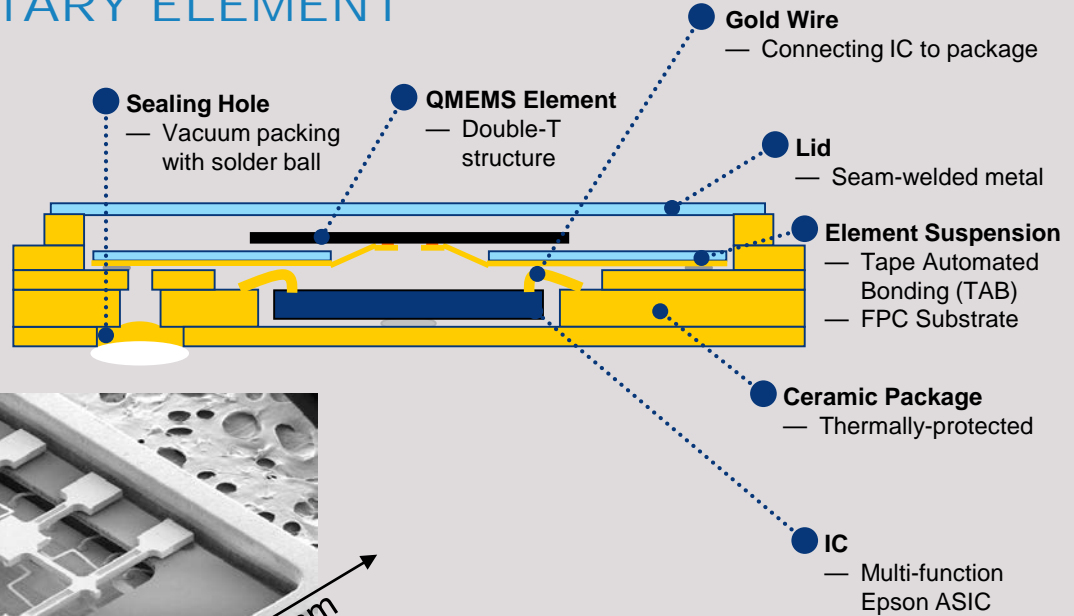


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SECTION II: GYRO TECHNOLOGY PROPRIETARY ELEMENT

Architecture
 Epson's QMEMS gyro
 components and
 housing:



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Proprietary Element

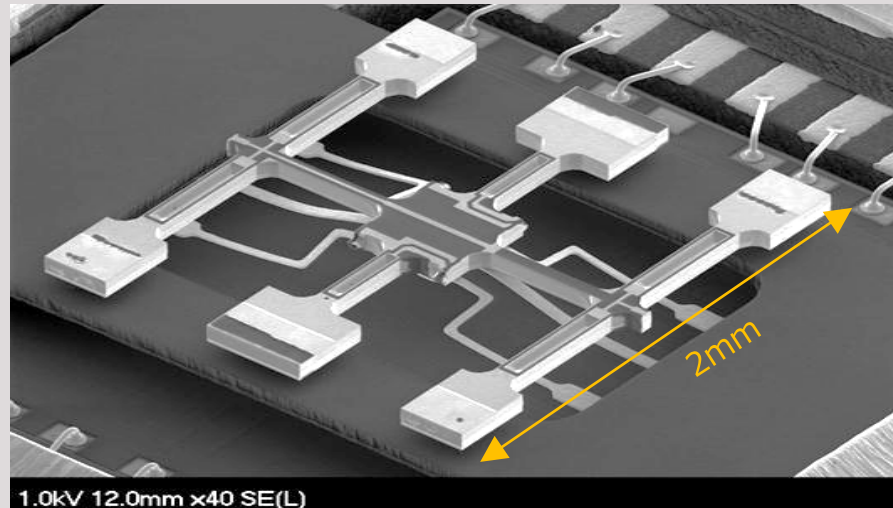
- Physical Structure
- **Architecture**
- Differentiation
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SECTION II: GYRO TECHNOLOGY PROPRIETARY ELEMENT

Architecture

Epson's QMEMS Double-T gyroscopes use a proprietary TAB mounting structure:

- Tape Automated Bonding.
- Provides significant shock & vibration isolation.



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Proprietary Element

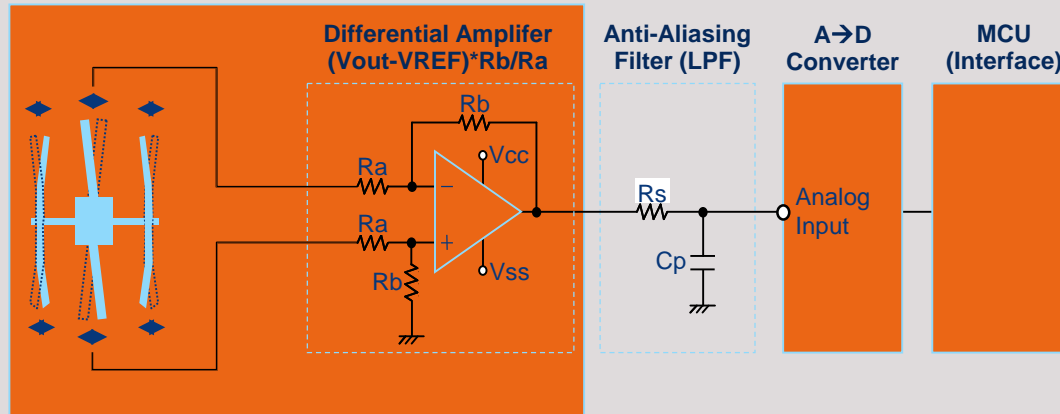
- Physical Structure
- **Architecture**
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- Noise Density

SECTION II: GYRO TECHNOLOGY PROPRIETARY ELEMENT

Architecture

Epson's QMEMS gyro electronics design advantages:

- Vibration and shock suppression due to differential amplification of two sensor arm signals.
- Amplification can be optimized for required gyro dynamic range.
- Individual control of anti-alias filter and A→D sampling rate.
- Intrinsic QMEMS sensor stability through temperature, including bias-drift and scale factor error.



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SECTION II: GYRO TECHNOLOGY PROPRIETARY ELEMENT

Differentiation

Epson's proprietary "Double-T" quartz MEMS gyroscopic sensor offers several advantages:

- Drive and detection arms are discrete but oscillate in the same plane.
 - No vibration is induced by the drive arms.
 - Signal-to-noise ratio is very high.
- Significantly lower noise than traditional H-type vibration gyros.
- Excellent rejection of vibration and shock.
- High stability over temperature.
- Very low power consumption.

| Gyroscope Element Structure | Epson Double-T | Tuning Fork | Silicon MEMS |
|-----------------------------|---------------------|-------------------------|---------------------|
| Q Value | ⊙ Q=30000 | ○ Q=10000 | △ Q=3000 |
| Element Support Method | ⊙ Point symmetry | △ Cantilever support | ○ Center support |
| Detecting Structure | ⊙ separation | △ No separation | △ No separation |

Ask the Experts – Part 1



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Ryan Dixon
Chief Engineer, SPAN
NovAtel

Moderator: Demoz Gebre-Egziabher

Poll #2

When considering the purchase of an INS solution, how important is the quality of the IMU in your decision?

- 1. Very important*
- 2. Important*
- 3. Somewhat important*
- 4. Not important*
- 5. Not sure*

EPSON IMU FEATURE SET



David Gaber
Marketing & Business
Development
Epson

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Proprietary Element

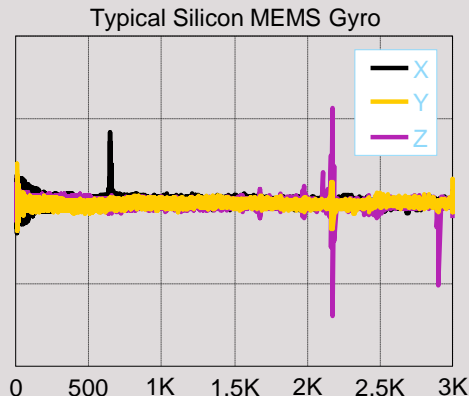
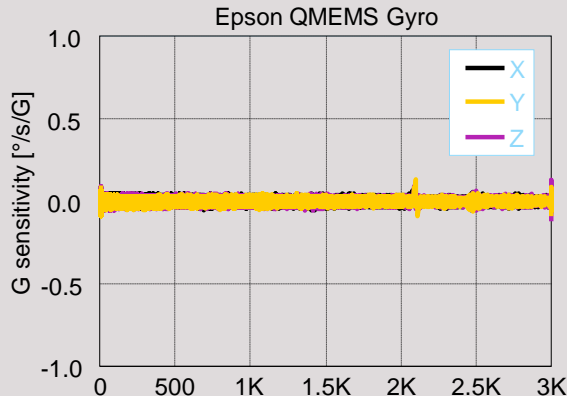
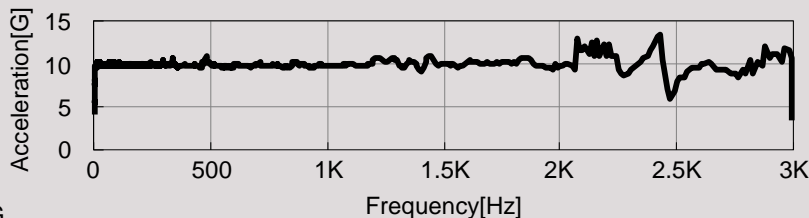
- Physical Structure
- Architecture
- Differentiation
- **Vibration Effects**
- Shock Effects
- Temperature Effects
- Noise Density

SECTION II: GYRO TECHNOLOGY PROPRIETARY ELEMENT

Vibration Effects

Epson's QMEMS gyros offer high resilience to induced vibration error:

Input motion
f=10~3000Hz
Acceleration 10G



Frequency[Hz]

- OVERVIEW
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Proprietary Element

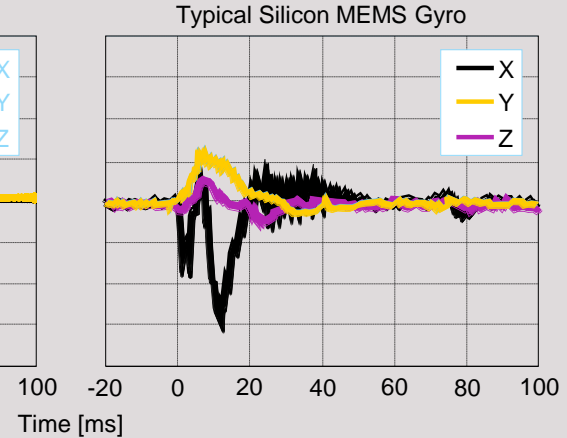
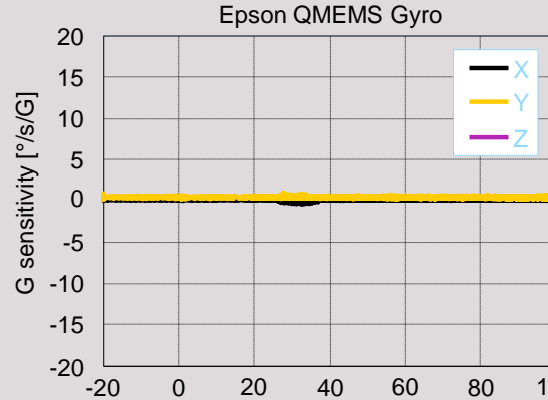
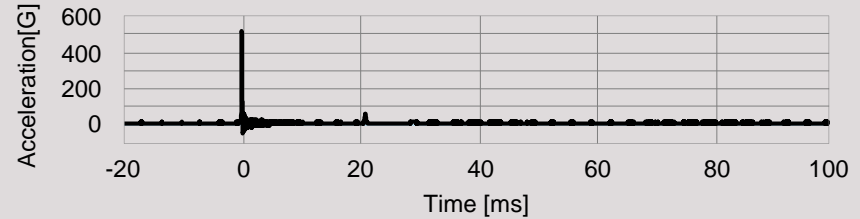
- Physical Structure
- Architecture
- Differentiation
- Vibration Effects
- **Shock Effects**
- Temperature Effects
- Noise Density

SECTION II: GYRO TECHNOLOGY PROPRIETARY ELEMENT

Shock Effects

Epson's QMEMS gyros offer high resilience to induced shock error:

Input motion
Acceleration
500G
1ms



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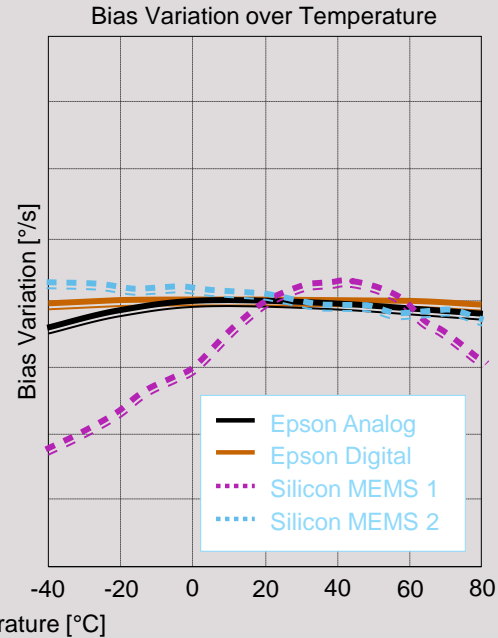
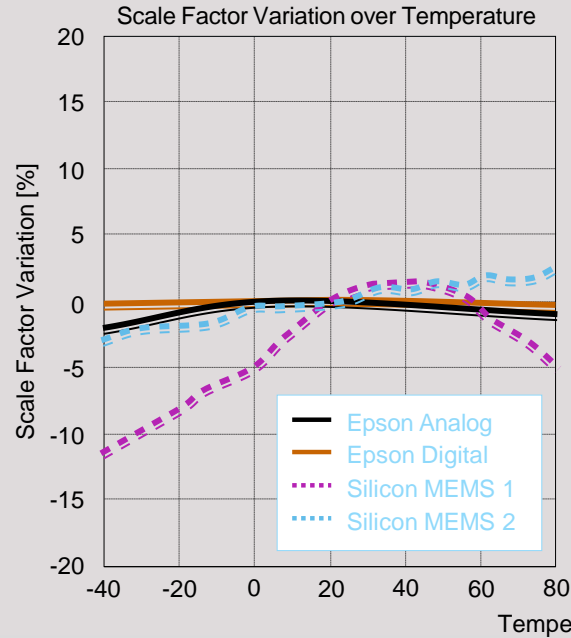
Proprietary Element

- Physical Structure
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- **Temperature Effects**
- Noise Density

SECTION II: GYRO TECHNOLOGY PROPRIETARY ELEMENT

Temperature Effects

Epson's QMEMS gyros offer high stability over temperature:



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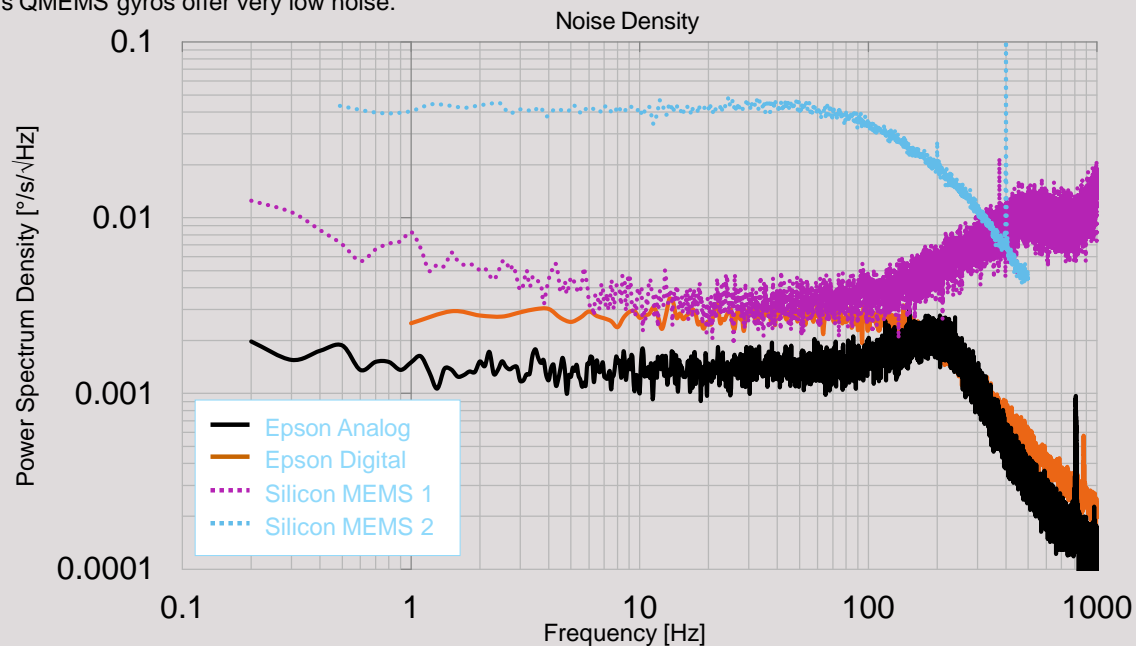
Proprietary Element

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SECTION II: GYRO TECHNOLOGY PROPRIETARY ELEMENT

Noise Density

Epson's QMEMS gyros offer very low noise:



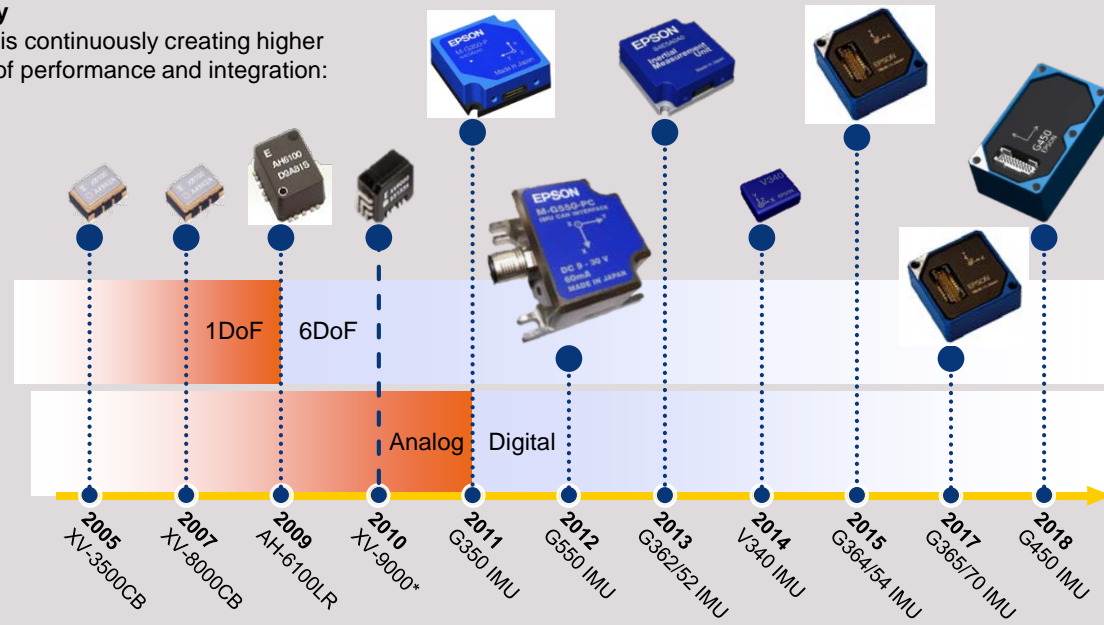
INTRODUCTION
 GYRO TECHNOLOGY
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20 Years of R&D
 — History
 — Performance

SECTION III: IMU EVOLUTION 20 YEARS OF R&D

History

Epson is continuously creating higher levels of performance and integration:



*The XV-9000 is a 1DoF gyroscopic sensor produced after Epson moved to 6DoF sensor types.

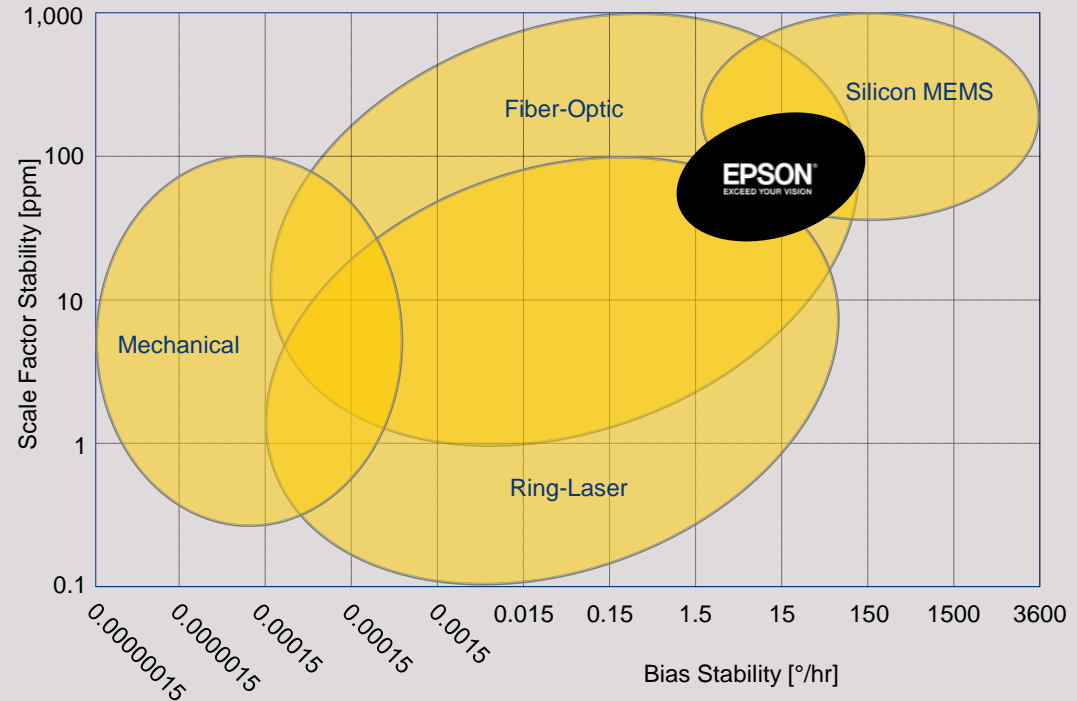
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20 Years of R&D
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SECTION III: IMU EVOLUTION 20 YEARS OF R&D

Performance

Epson IMUs deliver high performance without any export control restrictions:



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Flagship Products
— G3XX Series

SECTION IV: CURRENT PRODUCTS IMU & ACCELEROMETER

G364

Narrow dynamic range:

- Ideal for slow-moving vehicles
- Epson's highest performance IMU



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Increasing Performance
— G220 | G365 | G370
— G450

SECTION IV: NEW PRODUCTS

IMU

G220
G365
G370

| P/N | G220 | G320 | G354/364 | G365 | G370 |
|------------------------|------------------------------------|---------------|-----------------|----------------------------------|---------------------|
| Status | Sampling Now | MP | MP | Sampling Now | Sampling Now |
| Gyro | ±150dps | ±150dps | ±450/200dps | ±150/450dps | ±150/450dps |
| Bias Error [deg/sec,σ] | 0.1(z),0.5(x/y) | 0.5 | 0.1 | 0.1 | 0.1 |
| BIS[deg/hr] | <2(z),8(x/y) | 3.5 | 3/2.2 | < 1.8 | < 0.8 |
| ARW[deg/√hr] | <0.1(z),0.2(x/y) | 0.1 | 0.2/0.09 | 0.09 | 0.06 |
| Noise[deg/sec/√Hz] | 0.004 | 0.002 | 0.002 | 0.0015 | 0.001 |
| BW [Hz] | 50 | 200 | 200 | 500 | 500 |
| Accl | ±6G | ±5G | ±5/3G | ±6/10G | ±6/10G |
| Bias Error [mG,σ] | (TBD) | 15 | 5 | 3 | 2 |
| BIS [μG] | 100 | 100 | 70/50 | 10 | 7 |
| VRW [m/s/√hr] | 0.1 | 0.05 | 0.03/0.025 | 0.04 | 0.03 |
| Noise[μG/√Hz] | 200 | 100 | 60 | 70 | 50 |
| BW [Hz] | 50 | 200 | 200 | 500 | 500 |
| Data output | 16bit,< 1kSps | 32bit,< 2kSps | 32bit,< 2kSps | 32bit,< 2kSps | 32bit,< 2kSps |
| Attitude Output | | (N/A) | | Tilt Angle (up to 200sps) | |
| IF | UART/SPI (20-pin connector) | | | | |
| PKG | 24x24x10mm | | | | |
| Temp. Operation [°C] | -40to+85 | -40to+85 | -40to+85 | -40to+85 | -40to+85 |
| Calibration [°C] | -20to+70 | ↑ | ↑ | ↑ | ↑ |
| Power | 3.3V, 16mA | 3.3V, 18mA | 3.3V, 18mA | 3.3V, 18mA | 3.3V, 18mA |

INTRODUCTION
 GYRO TECHNOLOGY
 IMU EVOLUTION
 CURRENT PRODUCTS
NEW PRODUCTS

Increasing Performance

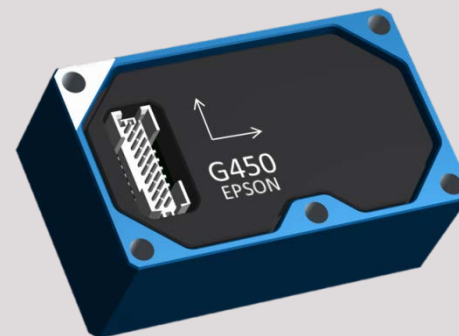
- G220 | G365 | G370
- G450

SECTION IV: NEW PRODUCTS

IMU

G450

| | Units | G450 (Draft) |
|---------------------------|---------------------------|--------------------|
| Gyro (Range) | deg/sec | 100 / 450 |
| SF Error | % (σ) | 0.05 |
| Bias Error | deg/sec (σ) | 0.1 |
| Bias Instability | deg/hr | 0.5 |
| ARW | deg/ $\sqrt{\text{hr}}$ | 0.03 |
| Accel (Range) | g | 3 / 6 |
| SF Error | % (σ) | 0.05 |
| Bias Error | mG (σ) | 2 |
| Bias Instability | mG | 0.01 |
| VRW | m/sec/ $\sqrt{\text{hr}}$ | 0.007 |
| Output Data Rate | Hz(max) | 1,000 |
| Resolution | bits | 16 / 32 |
| Interface | - | SPI/UART |
| External Trigger Accuracy | μsec | 100 |
| Attitude Output Function | - | Quaternion / Euler |
| Cal Temp Range | Deg.C | -40 to 85 |
| Power Supply | V / mA | 3.3 / (TBD) |
| Size | mm | (24 x 50 x 17) |

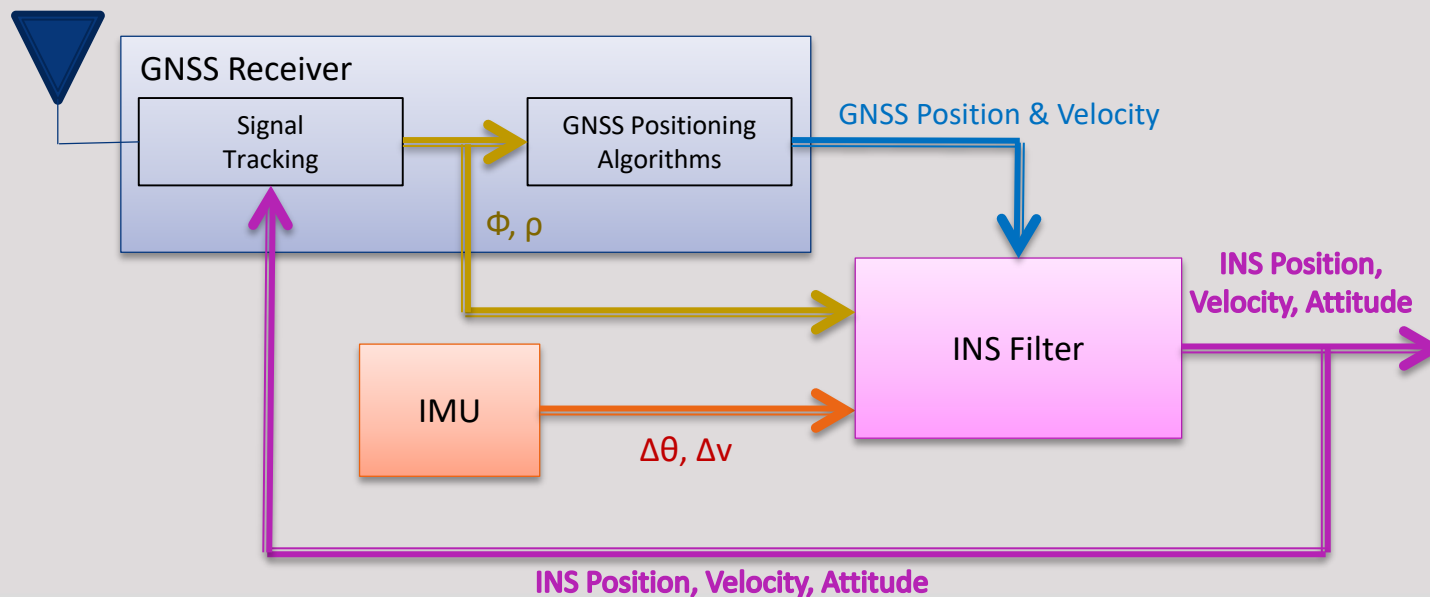


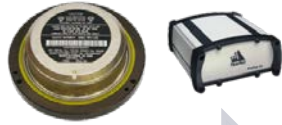
Performance Differentiation in a GNSS/INS Solution



Ryan Dixon
Chief Engineer, SPAN
NovAtel

- NovAtel's GNSS/INS product line
- SPAN = "Synchronized Position Attitude Navigation"
- Combining a range of IMU sensors with NovAtel GNSS receivers
- Deeply (Ultra-Tightly) coupled architecture





2003
OEM4 &
RLG /
Quartz



2005
FOG/
Quartz



2008
OEMV &
FOG /
MEMS



2012
OEM6 &
MEMS



2016
OEM7 &
MEMS

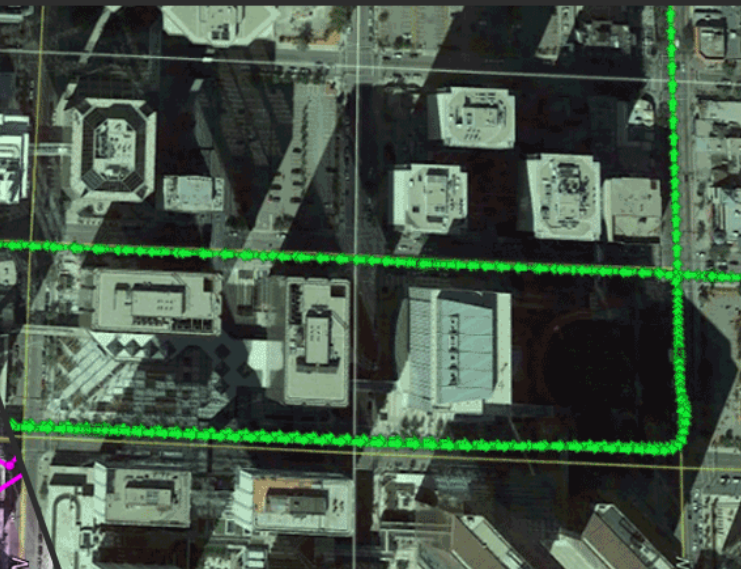


2017
Land
Vehicle
Tech
...

GNSS ONLY



WITH SPAN[®] Land Vehicle



HEXAGON
POSITIONING INTELLIGENCE



Designed to apply a variety of constraints to any fixed wheel land vehicle for a variety of applications

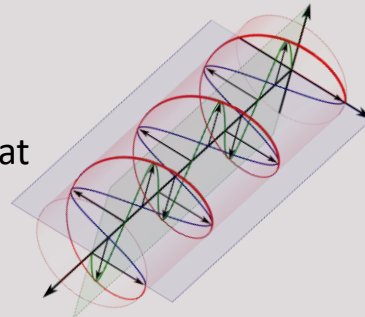


■ Vehicle velocity constraints

- Non-holonomic constraints
- Dead Reckoning
- If you don't have GNSS

■ Phase wind-up relative attitude

- Relative azimuth update method looking at the circular polarization of GPS signals
- Greatest improvements in low motion environments



Source: Wikipedia



■ Robust kinematic alignments

- Allow alignments as easily as possible. Do not force a specific alignment procedure
- Automatically detect forward or backwards start

Three test cases to examine:

- Urban Canyon
- Low Dynamics
- Extended GNSS Outage

Tests use GNSS and INS only. No aiding sensors used and only single antenna GNSS solution



PERFORMANCE¹

Channel Configuration

555 Channels

Signal Tracking

GPS L1 C/A, L1C, L2C, L2P, L5

GLONASS² L1 C/A, L2C, L2P,

L3, L5

BeiDou³ B1, B2, B3

Galileo⁴ E1, E5 AltBOC

E5a, E5b, E6

NavIC (IRNSS) L5

SBAS L1, L5

QZSS L1 C/A, L1C, L2C, L5, L6

L-Band up to 5 channels

GNSS Horizontal Position

Accuracy (RMS)

Single point L1 1.5 m

Single point L1/L2 1.2 m

NovAtel CORRECT™

» SBAS⁵ 60 cm

» DGPS 40 cm

» PPP⁶

 TerraStar-L 40 cm

 TerraStar-C 4 cm

» RTK 1 cm + 1 ppm

 Initial time <10 s

 Initial reliability >99.9%

IMU PERFORMANCE¹¹

Gyroscope Performance

Input range ±150 deg/s

Rate bias stability 3.5 deg/hr

Angular random walk

0.1 deg/√hr

Accelerometer Performance

Range ±5 g

Bias stability 0.1 mg

Velocity random walk

0.5 m/s/√hr

COMMUNICATION PORTS

1 RS-232 up to 460,800 bps

2 RS-232/RS-422 selectable

up to 460,800 bps

1 USB 2.0 (device) HS

1 USB 2.0 (host) HS

1 Ethernet 10/100 Mbps

1 CAN Bus 1 Mbps

3 Event inputs

3 Event outputs

1 Pulse Per Second output

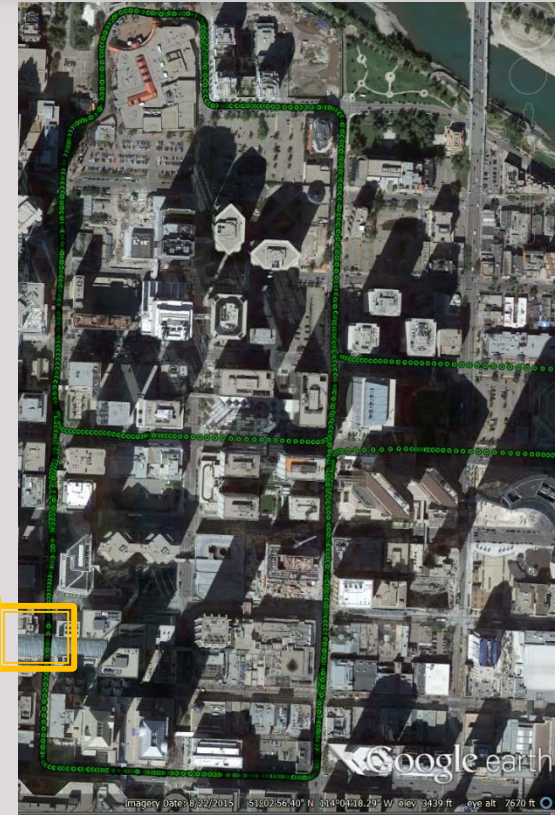
1 Quadrature Wheel Sensor

input

PHYSICAL AND ELECTRICAL

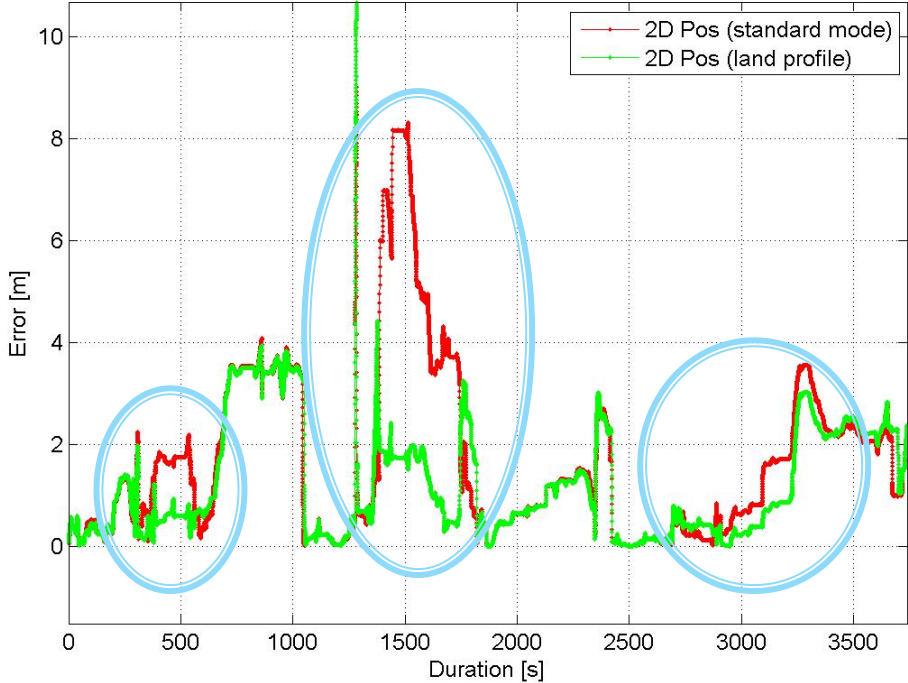
Test Case 1: Urban Canyon

- Downtown Calgary, Canada
- Difficult GNSS conditions
- Benefits largely from SPAN tightly-coupled architecture; use of partial GNSS information
- SPAN Land Vehicle technology aids during the most difficult periods

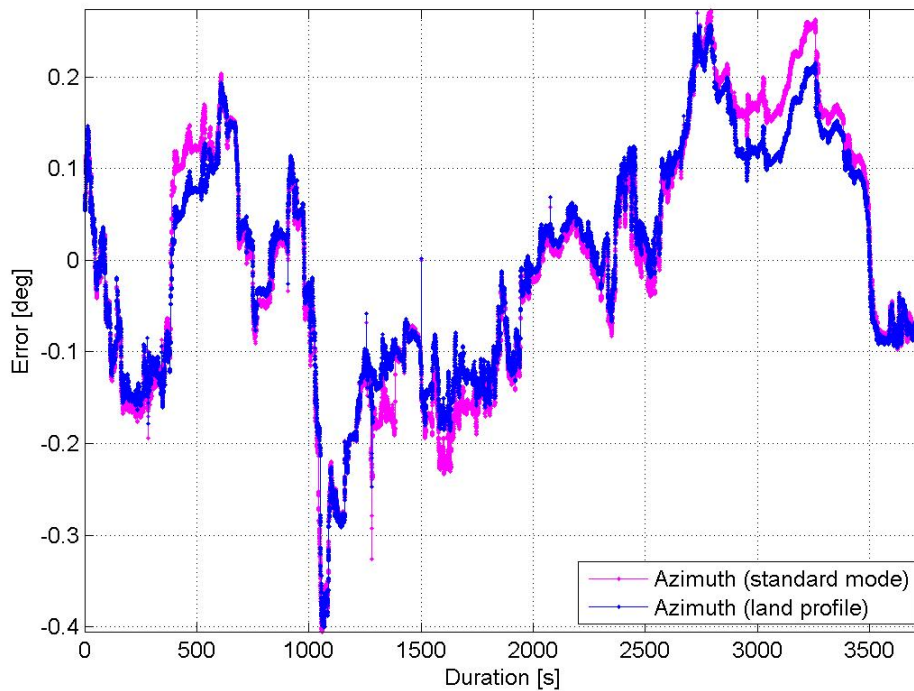


Source: Google Earth

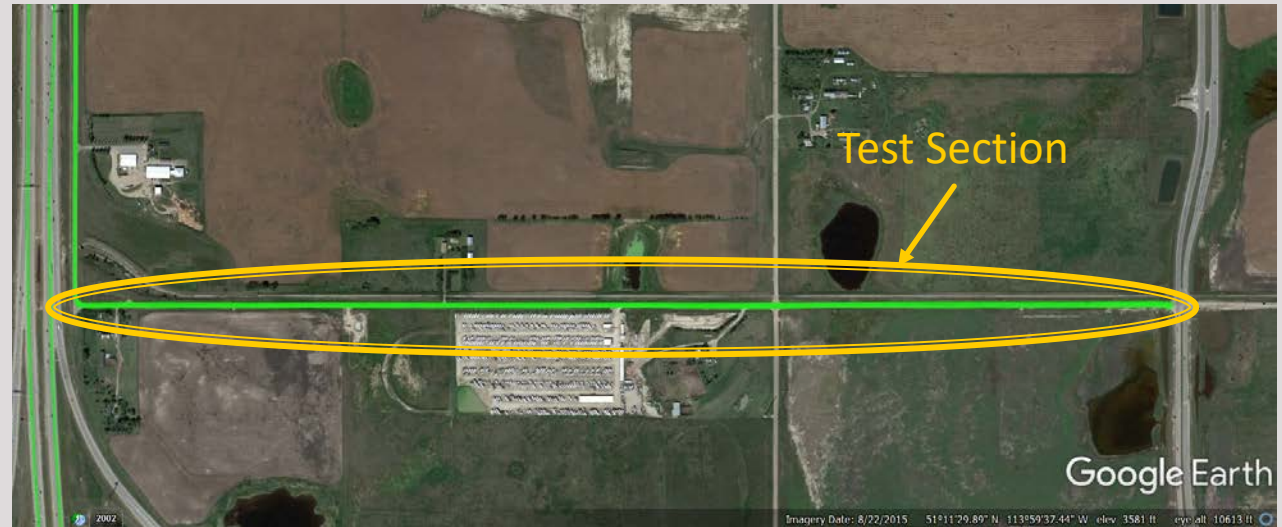
Urban Canyon Test
Epson G320 Position Error (standard vs land profiles)



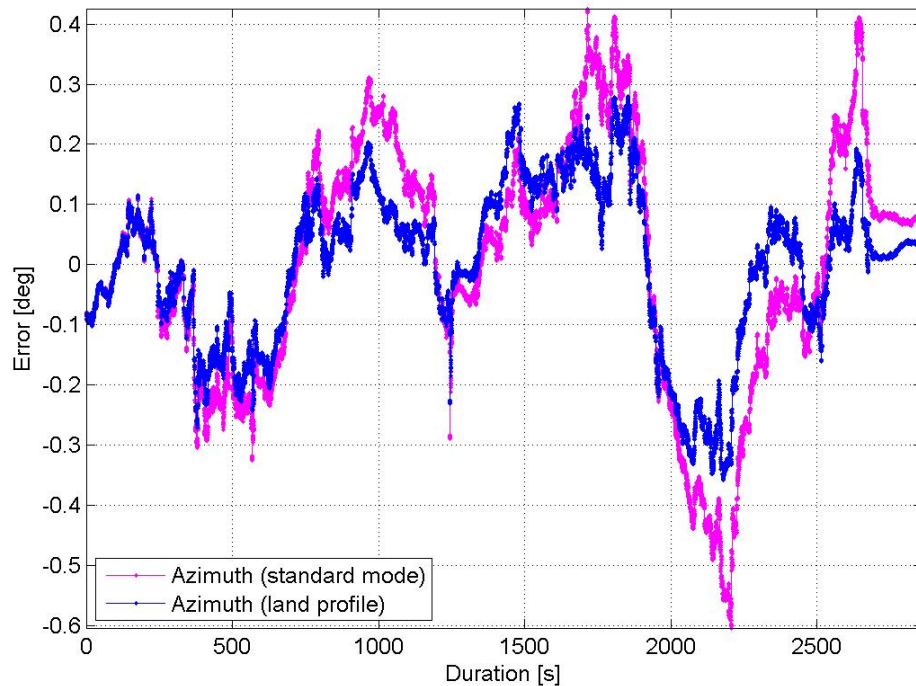
Urban Canyon Test
Epson G320 Attitude Error (standard vs land profiles)



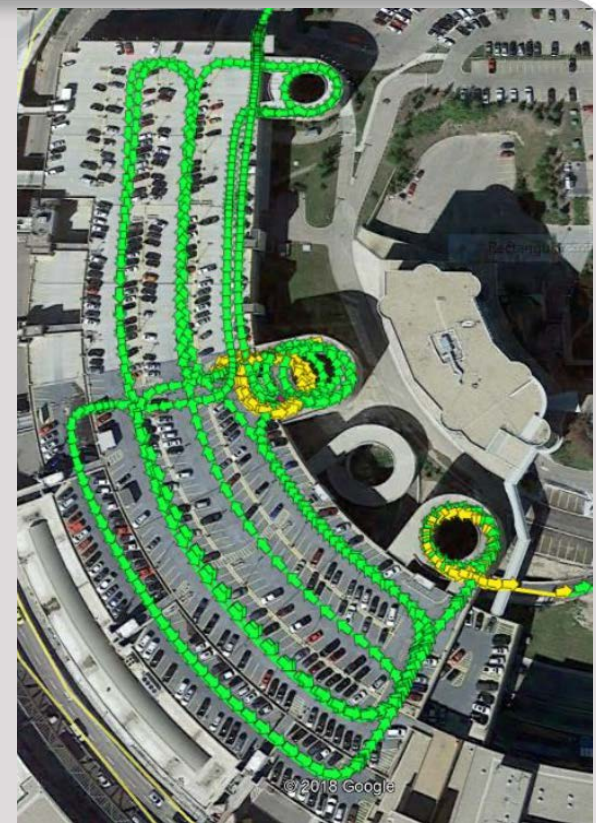
- Vehicle moving straight on 2 Km rural road at 10-15 Km/h
- Ideal GNSS conditions
- Difficult INS conditions; limited observable motion
- Benefits from vehicle motion constraints and phase windup attitude updates



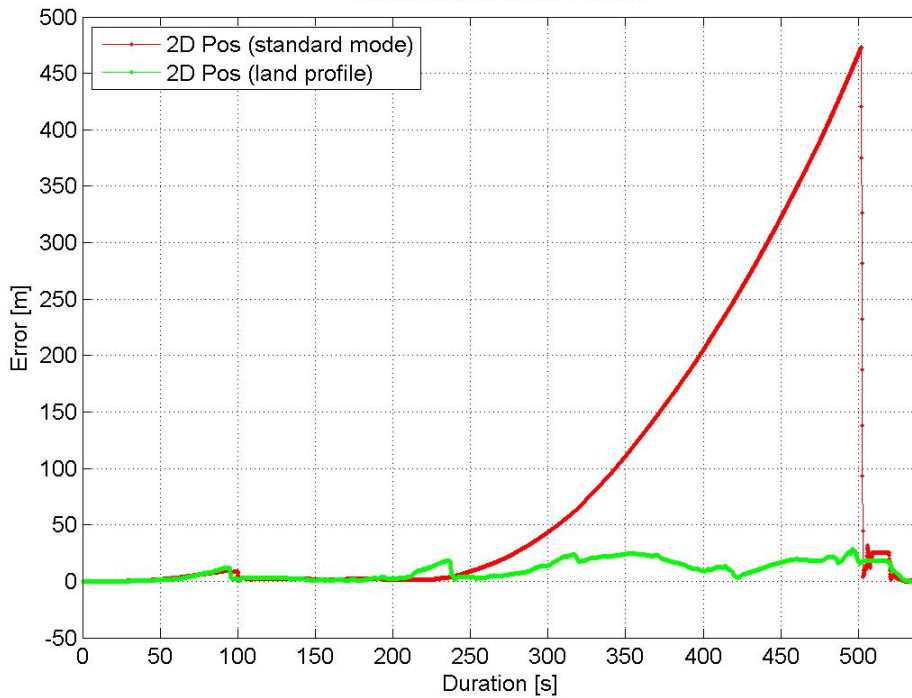
Low Dynamics Test
Epson G320 Attitude Error (standard vs land profiles)



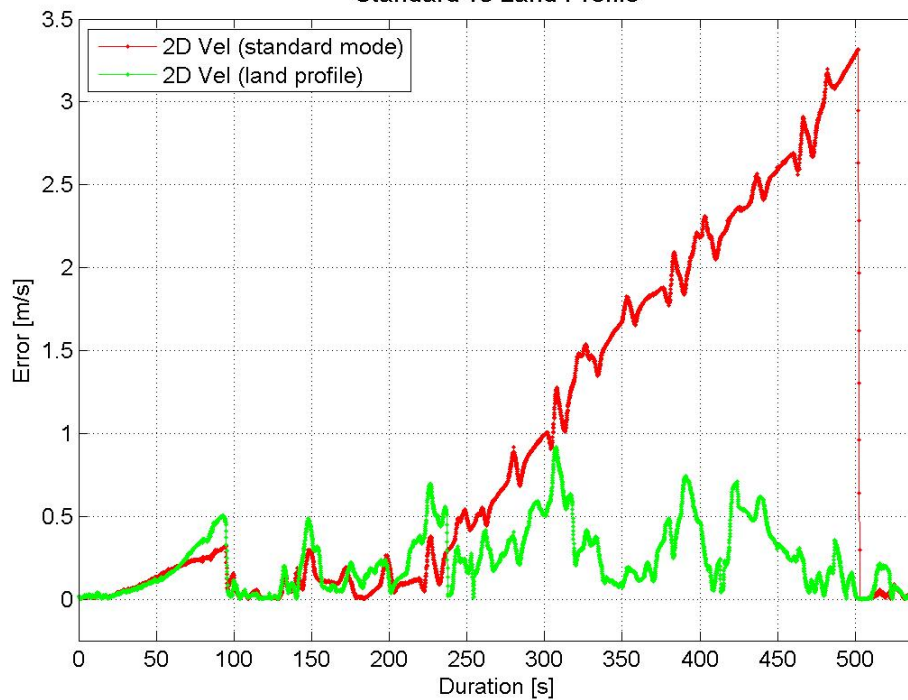
- Parking Garage – extended GNSS outage
- Relying on the propagated INS solution
- Performance driven by IMU and application of land vehicle constraints



YYC Parkade Test
Epson G320 Position Error
Standard vs Land Profile



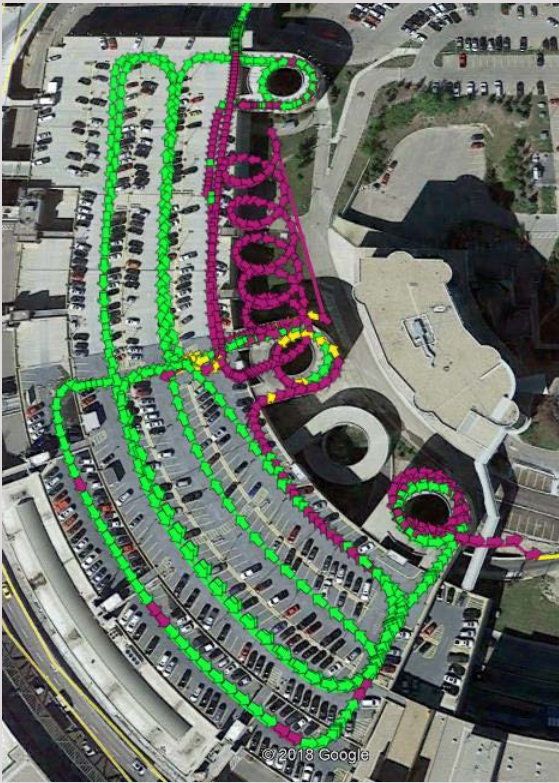
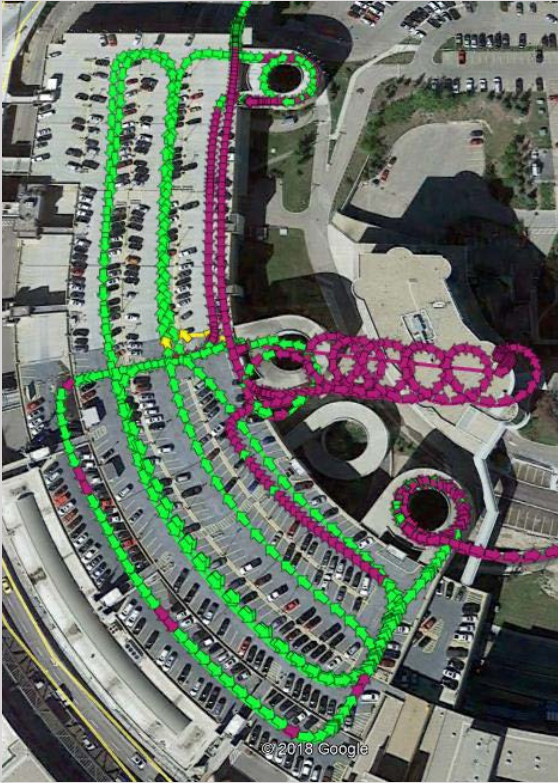
YYC Parkade Test
Epson G320 Velocity Error
Standard vs Land Profile

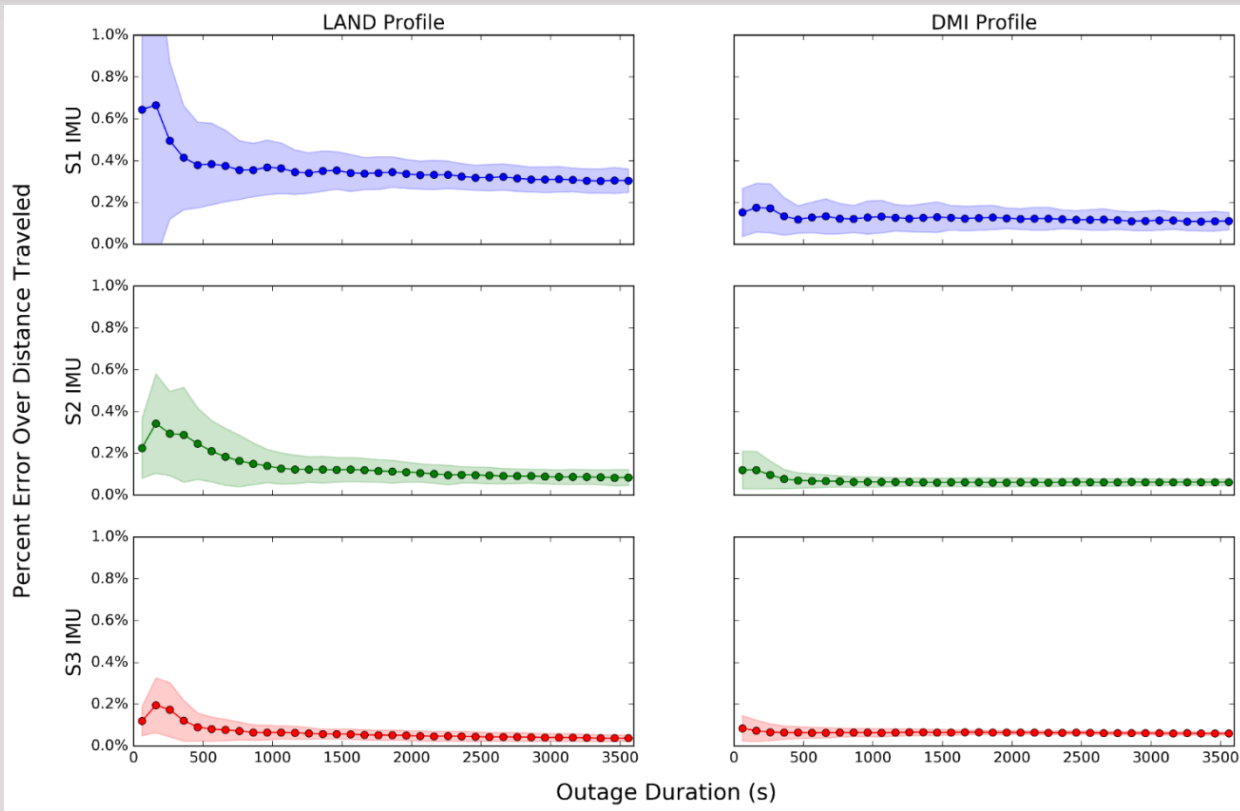


OEM6 SPAN

OEM7 Default SPAN

OEM7 with SPAN Land Vehicle





- Remarkable performance is now achievable with the latest MEMS IMUs but be informed
- IMU selection remains crucial
- INS algorithms differentiate performance
- System selection is very important
 - Know what the desired environment(s) are
 - Know the key performance metric(s)

Poll #3

In what harsh GNSS environments do you struggle to provide high accuracy positioning? (top three)

- A. Urban canyons and/or foliage*
- B. Jammed*
- C. Spoofed*
- D. Indoors*
- E. Tunnels/Underground/Pipeline*

SPAN Land Vehicle Performance Analysis Paper: [NovAtel's SPAN Land Vehicle Performance Analysis.](#)

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Ask the Experts – Part 2



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and Mechanics
University of Minnesota



Andrey Soloviev
Principal
QuNav



David Gaber
Marketing & Business
Development
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